

# Toward a complete metrologic solution for the mirrors for the Constellation-X spectroscopy x-ray telescope

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## ABSTRACT

We present an overview update of the metrologic approach to be employed for the segmented mirror fabrication for Constellation-X spectroscopy x-ray telescope. We compare results achieved to date with mission requirements. This is discussed in terms of inherent capability versus in-practice capability. We find that all the needed metrics for the mirrors are in hand but that they are currently limited by the mounting of the mirrors themselves.

**Keywords:** Optical metrology, x-ray optics, Constellation-X

## 1.0 INTRODUCTION

### 1.1 Constellation-X & Spectroscopy X-ray Telescope Overview

The Constellation-X mission<sup>1,2</sup> is a spectroscopic x-ray mission to be flown in 2017 as a complement to the current Chandra x-ray mission. By combining a higher effective area and greater spectral resolving power at the Iron K lines, this observatory will be a powerful tool for observing black holes, investigating dark matter, and elucidating galactic evolution.

The mission concept consists of a single spacecraft with four 1.3m, ~12.5 arcsec HPD soft x-ray telescopes (SXT) with 163 nested, azimuthally segmented shells of grazing incidence mirrors. In the current configuration, each SXT has shells approximately 0.4 mm thick and segmented to about 36° or 72° to keep the segment arc length to a maximum of about 40 cm. The substrate material is glass (Schott D263) coated with iridium. Details of the segment fabrication are described elsewhere.<sup>3</sup> Each telescope is segmented into 5 inner and 10 outer modules (each with many nested primary/secondary reflector pairs) to ease assembly. Each SXT focuses onto a microcalorimeter array for x-ray nondispersive spectroscopy.

### 1.2 Metrology requirements for the segmented x-ray mirrors for the SXT

The metrology requirements are based on the mission error budget presented previously.<sup>4,5</sup> The error budget, however, is not cast in terms of the metrology involved so some manipulation and equivalence adjustments are performed to recast the error budget in terms that are readily elucidated from the metrologic methods themselves. Metrology of the mandrels used for forming the thin glass mirrors presents its own unique challenges and is not presented here. Table 1 lists the salient error budget terms for the mirrors along with the metrology accuracy needed to determine those requirements. The old<sup>6</sup> and new metrology method for each term is also shown. This paper concentrates on the current methods and their performance.

### 1.3 Instrumental versus in-practice performance

For the thin glass mirrors, the measurement accuracy never reaches the instrumental accuracy. This can be separated into two primary causes inherent to the problem: self weight deformation and vibration. Reference 6 discusses both of these in terms of plate theory, so we will not repeat that discussion here.

Static self-weight deformation is common in all metrology but is made more severe in the case of the thin glass mirrors where the thickness to width aspect ratio exceeds 1/100. These self-weight deformations exceed or are of the order of the metrology requirements themselves. Thus, discussion of the mirror shape independent of its mount becomes superfluous. It proves necessary to solve both the mount and measurement problem simultaneously. We postpone specific mount discussions to the sections on the measurement methods with which they are associated.

Table 1: Summary of Constellation-X mirror metrology method evolution 2004-2007.

Error Term	Budgeted Error	Metrologic Requirement	Current Metrology Method	Old Metrology Method
Average Radius	100 $\mu\text{m}$	33 $\mu\text{m}$	CCMM	CCMM
Cone angle	30 arcsec	10 arcsec	Hartmann <sup>1</sup>	CCMM
Roundness error	5 $\mu\text{m}$ RMS	0.6 $\mu\text{m}$ RMS	Hartmann/ Null Lens	CCMM
$\Delta\Delta R$	0.5 $\mu\text{m}$ RMS	0.2 $\mu\text{m}$ RMS	Hartmann/ Null Lens	CCMM
Average Axial Sag	0.31 $\mu\text{m}$ P-V	0.1 $\mu\text{m}$ P-V	Null Lens	Fizeau Interferometry
Axial Sag Variation	0.11 $\mu\text{m}$ P-V	0.03 $\mu\text{m}$	Null Lens	Fizeau Interferometry
Axial Slope irregularity	2.4 arcsec RMS	0.8 arcsec RMS	Null Lens	Fizeau Interferometry
Microroughness <sup>2</sup>	0.5 nm,RMS	0.2 nm,RMS	Mireau Interferometry	Mireau Interferometry

<sup>1</sup>The Hartmann test measures focal length so we must combine with the CCMM radius to calculate the cone angle

<sup>2</sup>Microroughness defined to be over the spatial period band from 1 mm to 1  $\mu\text{m}$

The second issue in thin mirror metrology is dynamic stability. Again this is complicated by the mounting method. Inherently, however, vibrational displacements resultant from (unsuppressed) environmental factors can be of similar or larger magnitude than the metrology uncertainty requirements, further complicating the measurements. Thus, the mount must reduce the excursions of the part or the metrology method must be insensitive to the vibrations. We have adopted both vibration suppression and less vibration sensitive metrology in the various scenarios discussed below.

Another aspect of the dynamic stability is one of timescales. It is important that the timescales of the mirror and mount dynamics be within the capture range of the metrology method. For the mirror metrology discussed here, there are several timescales involved. The mirror vibrations are a tenth of a second or less, air currents (if applicable) are a few seconds to a minute,. The stages used for alignment can drift over a few hours (often vibration driven), and the thermal drift of the apparatus ranges from a few to many hours. Because of these timescales, we have found that metrologic methods where a single measurement can be performed in much less than the air current and stage drift times are the most repeatable. This is true for both the Hartmann test §2.1 and the strobe Fizeau-null lens measurement §2.3.

It proves to be necessary to address the dynamic stability first. Once the dynamics have been dealt with, the static deformations can be determined unambiguously. Then the mount can be modified as needed to reduce the static deformation to a level acceptable for the error budget term(s) in question.

We will mention that the results presented here are still influenced by both self-weight deformation and dynamic influences. Of note, however, is that the dynamic influences are now being addressed well enough that they are no longer the dominant source of error in the measurements. Table 2 shows the error budget terms, metrology method used to measure them, and the in-practice metrology uncertainty achieved to date. The in-practice uncertainty includes the influence of the dynamic effects.

## 2. METROLOGY METHODS

### 2.1 Grazing-incidence Hartmann Test

The Hartmann test is well known in near-normal-incidence optics.<sup>7</sup> As one goes to grazing incidence, the mirror shrinks to essentially a single circular zone of constant radius. Thus, a grazing-incidence Hartmann test samples the entire radial extent of the mirror but only a small sector azimuthally.

Table 2: Summary of error budget terms and current metrology uncertainties. Note these in-practice uncertainties include (and are dominated by) the mirror fixturing.

Error Term	Budgeted Error	Metrology Method	In-Practice Uncertainty	Notes
Average Radius	100 $\mu\text{m}$	CCMM	20 $\mu\text{m}$	2
Cone angle	30 arcsec	Hartmann	10 arcsec	1
Roundness error	5 $\mu\text{m}$ RMS	Hartmann/ Null Lens	0.5/0.06 $\mu\text{m}$ RMS	1/1
$\Delta\Delta R$	0.5 $\mu\text{m}$ RMS	Hartmann/ Null Lens	0.4/0.04 $\mu\text{m}$ RMS	1/2
Average Axial Sag	0.31 $\mu\text{m}$ P-V	Null Lens	0.004 $\mu\text{m}$ P-V	2
Axial Sag Variation	0.11 $\mu\text{m}$ P-V	Null Lens	0.08 $\mu\text{m}$ P-V	2
Axial Slope irregularity	2.4 arcsec RMS	Null Lens	0.05 arcsec	2
Microroughness	0.5 nm,RMS	Mireau Interferometer	0.2 nm RMS	

1. In practice value estimated from a small sample

2. In practice determined from a statistically significant sample

In practice, we employ slits rather than holes as is typical in a normal-incidence Hartmann test (see Figure 1). This assures that the mirror is fully illuminated radially and any diffraction effects are from the mirror's limited radial extent rather than the mask itself. The slits are 2 mm wide. Thus, the radial blur and azimuthal blur in the test are a combination of the mirror performance and diffraction.

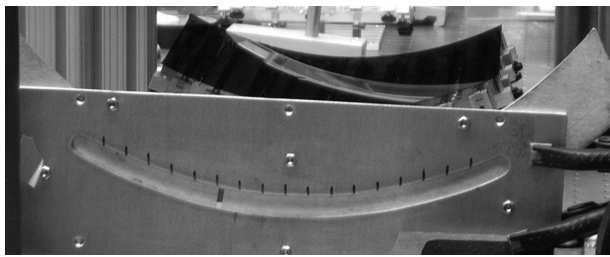


Figure 1: Grazing-incidence Hartmann mask (foreground) and mirror under test (dark object in middle ground). The mirror is on a self-leveling spring-based low distortion mount.<sup>8</sup>

We currently employ 15 slit positions across a 50 degree azimuthal span. This translates to about a 15 mm center-to-center spacing on the mirror under test for the nominal average radius of 244.5 mm of the prototype mirror segments. A He-Ne laser is spatially filtered and collimated using a high quality off-axis parabola. The mounted mirror segments under test are placed in the collimated beam on a six degree of freedom (6-DOF) stage and the beam folded (one or three times, depending on the focal length of the mirror under test) with high quality mirrors until the light comes to focus on an Apogee Instruments, Inc. AP47P CCD detector. A neutral density filter is placed in front of the detector to eliminate any stray light from the room.

The detector is initially placed at the nominal focal length (5.6 m for the secondary or 16.8 m for the primary) and the mirror adjusted using the 6-DOF until a best image is formed. The data are a set of centroids versus azimuthal angle as shown in Fig. 2. This information is fed into a program that determines, via a least-squares fit, the lowest-order positioning errors relative to the collimated beam. From these data the average cone angle can be calculated when combined with the average radius. From the residual of the fit, roundness and  $\Delta\Delta R$  for the part under test can be determined.

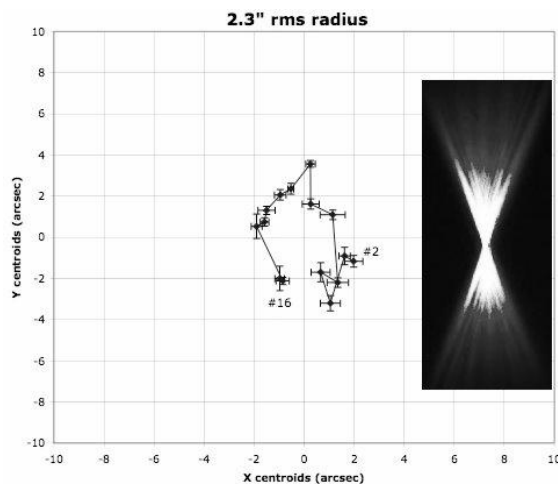


Figure 2: Example of a grazing-incidence Hartmann diagram for a primary mirror. The full-aperture image is shown in the inset. The rigid-body aberrations of defocus and coma are apparent, along with smaller residual errors in the part itself.

In practice the performance of this test is limited by the mounting of the mirror segment. At the present time, the mirror is laid horizontally on a self-leveling spring-based mount<sup>8</sup> seen in Figure 1 and this is mounted to the 6-DOF stage. The self-leveling mount relies on the gravity vector being nearly vertical along the centerline of the part so the adjustments made with the 6-DOF stage result in the gravity vector varying slightly from the original configuration which, in turn, results in the part moving in the mount in an undesirable way (not the way one wishes it to move

given the 6-DOF adjustment). This is prevented by a light tack bond with Optocast® 3415 at four points at the periphery of the mirror. The bonding process is monitored with the Hartmann apparatus itself to minimize the (local) distortion introduced. Results from ten tack bonding tests on secondary mirrors show no change in Hartmann RMS before and after bonding within experimental errors; results from five cyclic bondings on primary mirrors show a 0.8 arcsec RMS increase after tack bonding.

This technique is still under development as of this writing so the figures quoted in Table II are estimates based on the data sets available at this time.

## 2.2 Cylindrical coordinate measuring machine (CCMM)

We employ a custom designed non-contact coordinate measuring machine in a cylindrical geometry for determination of the large-scale geometry of the mirrors (Figure 3).<sup>6</sup> This CMM employs a confocal optical probe developed by Stil SA based on the variation of longitudinal chromatic aberration. A spectrum is obtained from the illuminated spot on the sample. This spectrum can be analyzed to obtain a precise distance relative to the calibrated probe. There is a significant trade-off between measurement range and precision. Our version uses Stil's highest sensitivity probe, with a quoted RMS distance uncertainty of approximately 2nm and about 20 microns total accessible range. It employs precision stages for radial position, vertical position, and azimuthal position with a measured radial noise of about 40 nm RMS. This is less than the goal for the instrument but still exceeding other commercial non-contact probes. A calibration flat is mounted at a separate azimuthal position adjacent to the test part. This allows calibration (linearization) of the portion of the stage travel of interest for both the radial and vertical stages.

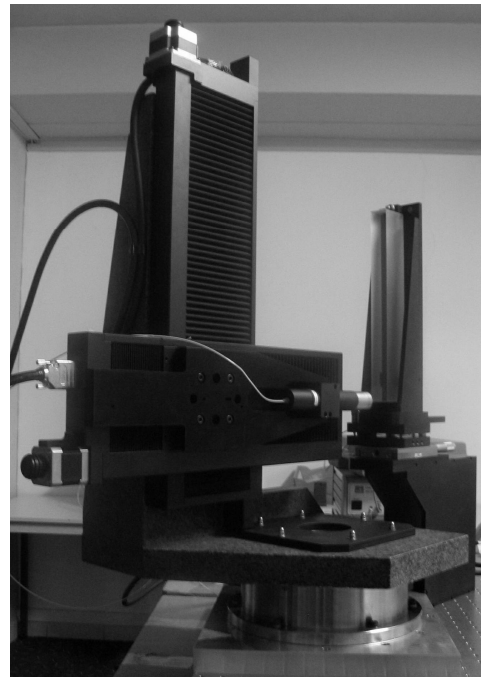


Figure 3: Non-contact Cylindrical Coordinate Measuring Machine (CCMM). The optical probe is attached to precision rotation, radial and z linear stages. The calibration flat is shown on the right but the part under test is not shown in this photograph.

The high data rate of the optical probe (2 kHz) allows dynamic studies of individual points on the mirrors. This compliments the interferometric measurements (§2.3) and can be employed in the verification of the resonance frequencies of the mirrors in a variety of fixtures.

The performance of the system is limited in practice by three factors: mirror vibration, mirror alignment stage movement, and thermal drift. The latter two factors are a result of the measurement time required for adequate sampling of an entire mirror. Each point requires sufficient samples to average over the vibrations at that point. Thus, the measurement of an entire mirror takes several hours to complete. In principle, the instrument can measure almost all the mirror parameters in Table 2. To accomplish this to the needed precision, however, the drift allowable is only tens of nm in the stages over the measurement time. In addition, the thermal stability needs to be of the same order over the same timeframe. As of now, we have not been able to achieve the combination of mount and thermal stability to measure all parameters with the required precision.

There is an additional factor that affects the utility of the CCMM. There is a mathematical degeneracy between cone angle and tilt angle of the segmented mirror relative to its axis. This indeterminacy is a function of the azimuthal span of the mirror and disappears for a full 360-degree shell.

In spite of these challenges, however, the achieved precision is sufficient to determine the average radius of the parts. The precision for this parameter has been found to be about 20  $\mu\text{m}$  (worst case), below the requirement of Table 1. The other parameters extracted from the CCMM data serve as double-checks for the parameters extracted from the Hartmann measurements.

### 2.3 Refractive null lens and strobe Fizeau interferometer

The combination of a refractive null lens and strobe interferometer covers the figure of the mirrors in the spatial period band from 200 mm to 1 mm. This normal incidence metrologic technique enables a much greater spatial fidelity of the mirror surface than practical with the grazing-incidence Hartmann technique or CCMM. Although the null lens is compatible with any commercial Fizeau interferometer with a sufficiently large collimated beam (>225 mm diameter here), we employ a 4D Technologies FizCam1500® with a 250 mm aperture. This strobe interferometer typically operates at a 0.2 msec staring time and a frame rate of 15 frames per second when performing our measurements. This limits the sampling of vibrational frequencies to only 7.5 Hz. Figure 4 illustrates the null lens and strobe Fizeau in use.

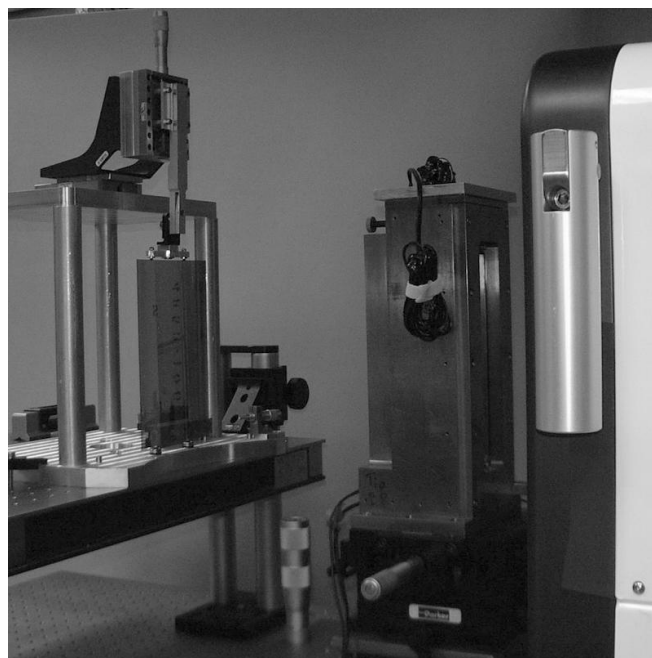


Figure 4: Cylindrical null lens and strobe Fizeau interferometer measuring mirror. Interferometer is to the far right (in the foreground); the cylindrical null lens is in the right center, and the part under test and metrology mount are on the left.

The null lens is cylindrical in geometry and its design has been discussed previously.<sup>9</sup> Of note here is that the refractive triplet cylindrical lens is capable of accommodating the cone angles of any of the telescope's mirrors. This has the advantage over other methods like diffractive optics because only a single lens is needed -- saving time and money but with the distinct disadvantage that the cone angle is not determined. A way of circumventing this is discussed in an allied paper.<sup>10</sup>

The calibration of the null lens and interferometer flat is performed in the usual way for interferometric systems. Where interferometry on thin glass mirrors deviates from

conventional interferometry is that good reproducibility requires extraordinary effort to achieve. As discussed above, the causes of non-reproducibility in our case can be divided into two categories: dynamic and static.

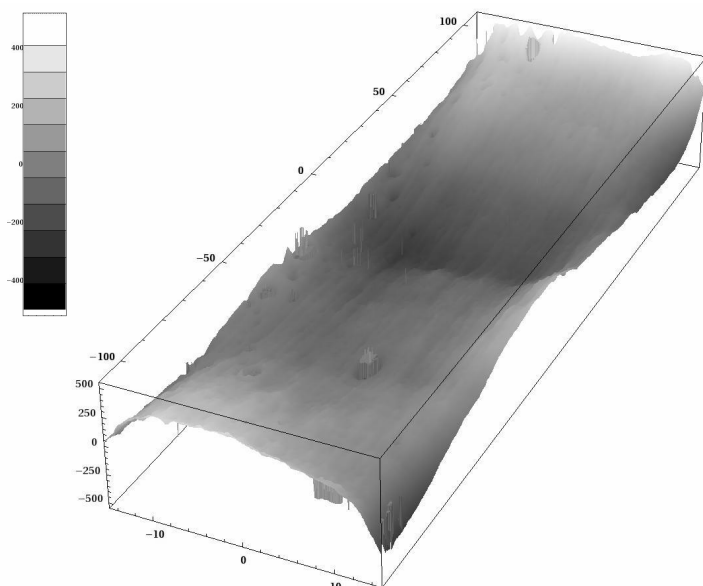


Figure 5(a): Typical surface map obtained from the average of 100 such maps taken with the set-up shown in Fig. 4. Axial scale ranges from -100 to 100 mm and the azimuthal scale is -15 to 15 degrees in the figure to the left.

The dynamic variations are due to vibration as discussed in §1.3 and air currents. We reduce air currents with curtains to create a static air environment and eliminating heat sources within the curtains (only the laser in the interferometer remains a source of any significance). This approach seems to be adequate to our purposes.

For vibrations we take two complimentary tacks. The first is the conventional metrologic approach of isolating environmental vibrations (vibration isolation tables and the like). We have found, however, that with reasonable effort, this approach

is insufficient for the vibrations of our mirrors. Our thin mirror's vibrational amplitudes are large even for small vibrational impulses. Thus, the second approach we employ is to measure the mirror while vibrating and average out the vibrations. We take advantage of the law of large numbers for this solution because the environmental sources of vibration are random in time, amplitude and frequency content.

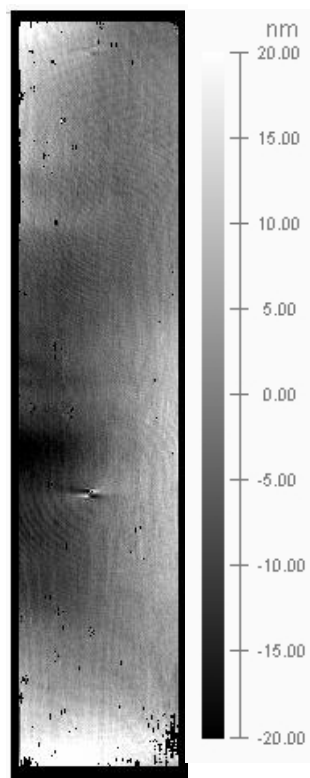


Figure 5(b): Difference plot between two subsequent averages for 100% of the null lens clear aperture. RMS error is 5.9 nm. Note the dominant residual seen is probably due to air currents

Figure 5(a) shows a typical average result whereas 5(b) shows the difference between two such averages with 5 nm RMS difference indicating good averaging by this combined approach.

The static portion of the non-reproducibility dominates the overall non-reproducibility after the averaging process is implemented and it is governed by fixturing. We employ a kinematic mount with the part in a vertical orientation (see Figure 4). The vertical orientation was chosen because it is the minimum deflection orientation since the self-weight deflection is due to column buckling. Because the part is actually a sagged cone rather than a cylindrical shell, however, only the centerline of the mirror is truly vertical during measurement. Thus, the periphery of the mirror sees a slightly greater fraction of the gravitation force normal to its surface than the centerline. There is a unique tilt angle that minimizes the peak-to-valley variation across the part that is slightly off the centerline vertical but the centerline vertical geometry is used in practice because it is easier to find precisely and repeatably.

The mount consists of bearings and fulcrums and is somewhat akin to a two-dimensional Whiffle tree. The bearings accommodate the possibility of locally varying slopes along the perimeter of the mirror without distorting the mirror. In the limit, it reduces to a

three-point kinematic mount similar to that discussed in reference 6. The mount is discussed in greater detail in an allied publication.<sup>10</sup>

The net variation caused by repeated mountings and dismountings is much greater than that from the residual uncertainty resultant from vibration when averaging is performed. The combined variability from all causes has been measured to be about 25 nm RMS. This translates to 0.08  $\mu\text{m}$  P-V (worst case) for the axial sag deviation (see Table II). This is below the requirement but still a bit above the metrologic goal (see Table I). Further optimization of the mount should allow us to achieve the metrologic goals on all the quantities measured with the null-lens-strobe-interferometer combination in the near future.

## 2.4 Mireau interferometry

We employ a Mireau interferometer to measure the mirrors at spatial periods 1 mm to 1  $\mu\text{m}$ . We employ two models (a Zygo NewView® and an ADE Phase Shift MicroXam®). The mirrors are sampled on a 5 x 5 grid and assumed statistically stationary in this spatial period regime. The gravity distortions appear only as tilt in this spatial period band so fixturing is greatly simplified. Performance of both instruments is comparable and we are regularly achieving the performance expected from the manufacturers' specifications.

## 3. SUMMARY AND CONCLUSIONS

We presented the error budget terms and associated metrologic goals for the Constellation-X mirror segments. We also presented the current metrology used to meet the particular requirements for those terms on these next-generation ultra-lightweight segmented x-ray optics. The principal new metrology developments include a refractive null lens and strobe interferometer, a vertical kinematic fixturing scheme to hold the optics for optical metrology with improved vibrational characteristics, a low-distortion self-leveling horizontal mount, and a grazing-incidence version of the classic Hartmann test for low order figure determination. The complete metrology set is at or near the metrology goals for the mission and is limited by the mounting of the mirrors. Further refinement of the mirror fixturing should put all the measurements comfortably in the regime of meeting the metrology uncertainty goals in the near future.

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